



POLITECNICO
MILANO 1863

CISLUNAR DEPARTURE EXPLOITATION FOR PLANETARY DEFENSE MISSION DESIGN

7° IAA Planetary Defense Conference – PDC 21

A. Pasquale, M. Lavagna, F. Renk

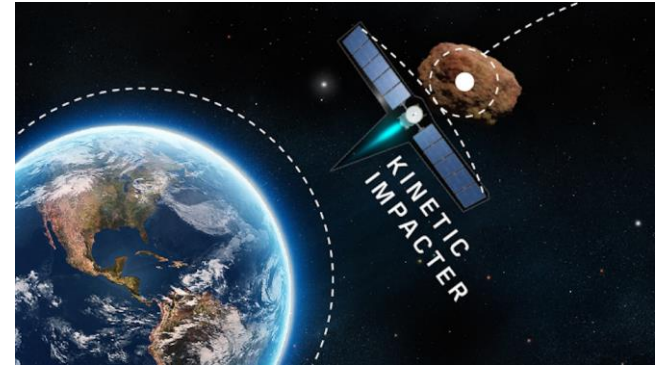


Context, Motivation & Objectives

International Space Exploration Coordination Group (ISECG) agencies increased interest in **cislunar** space.



Lunar Gateway (LOP-G)



Increasing concerns about **planetary defense** issues:

- Early **detection** of potentially hazardous objects;
- **Tracking** and characterization of PHOs;
- Issue **warnings** of the possible effects of potential impacts;
- Study strategies and technologies for **mitigating** PHO.

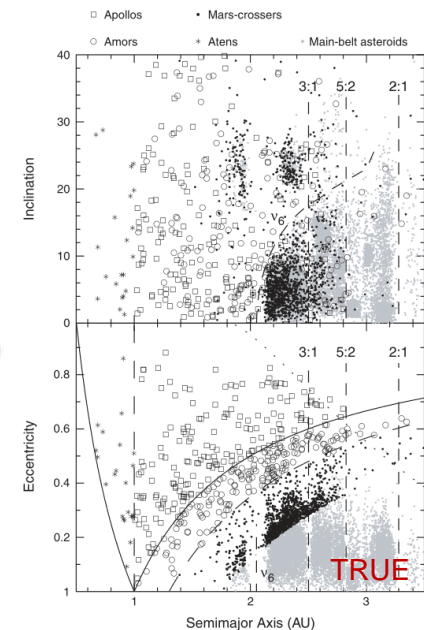
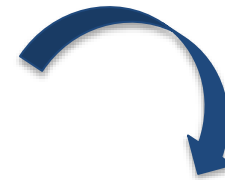
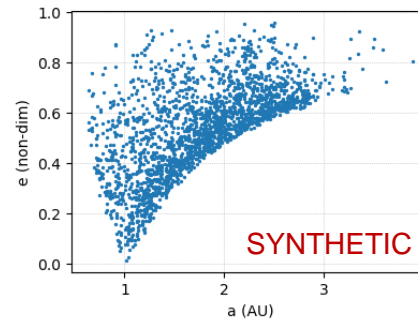
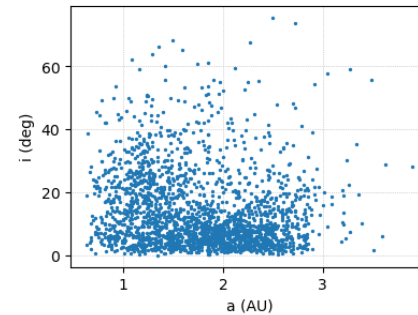
Preliminary assess the feasibility to exploit the cislunar space as a departing point for planetary defense missions.

NEA/PHA Modelling

- Generation of an arbitrary-sized NEO/PHO **synthetic** distribution
- Extraction of **tracked** object database (*merge NASA JPL Small Body and Minor Planet Center databases*).



Target region defined.



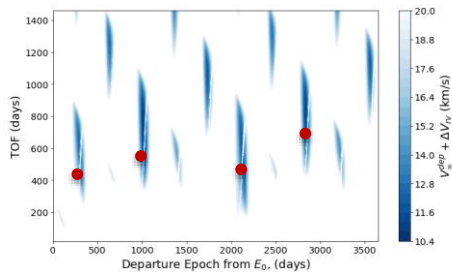
Mission Scenario Definition

Two-step optimization

RENDEZVOUS

The **total transfer energy** is minimized, i.e.

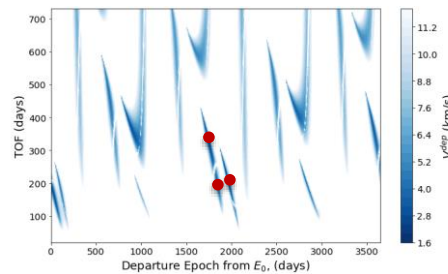
$$\min_{E_0, TOF} \sqrt{C_3} + \Delta V_{rv}$$



FLYBY

1. **Departure energy** minimization,

$$\min_{E_0, TOF} \sqrt{C_3}$$



2. Encounter **relative velocity** constrained minimization

$$\min |U| \quad s.t. V_\infty \in [0, cV_\infty^1]$$

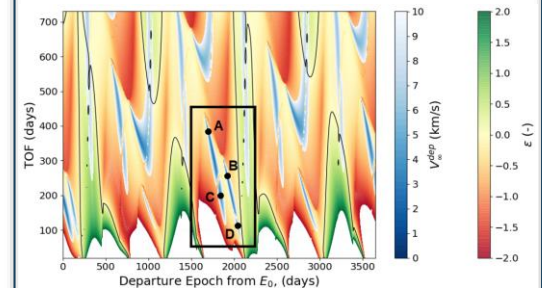
KINETIC IMPACT

1. **Departure energy** minimization,

$$\min_{E_0, TOF} \sqrt{C_3}$$

2. **Impact efficiency** maximization,

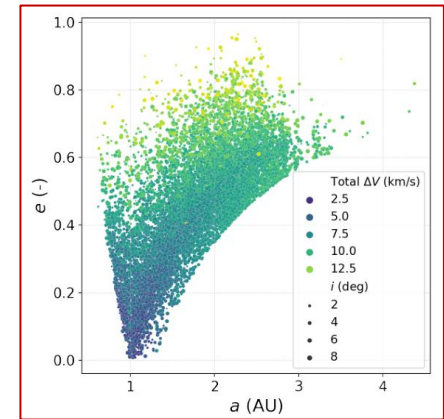
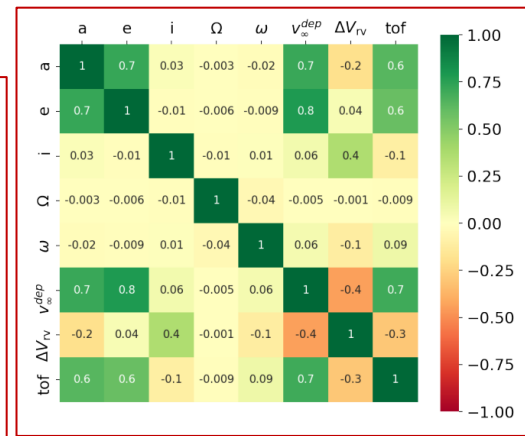
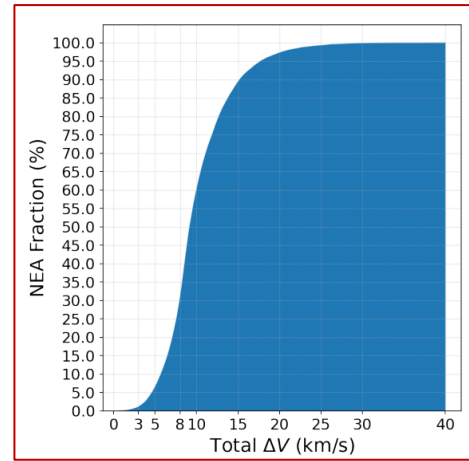
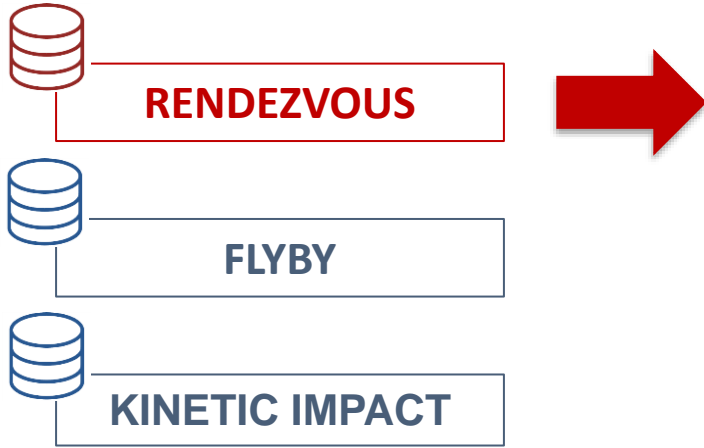
$$\min |1/\varepsilon| \quad s.t. V_\infty \in [0, cV_\infty^1]$$



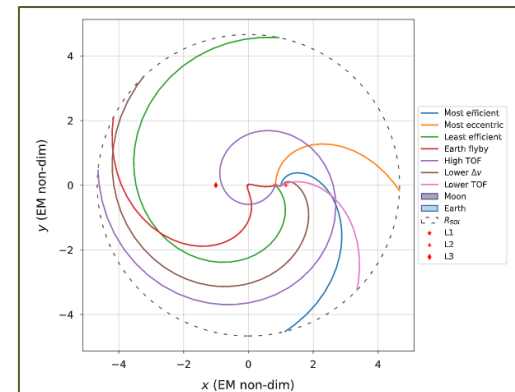
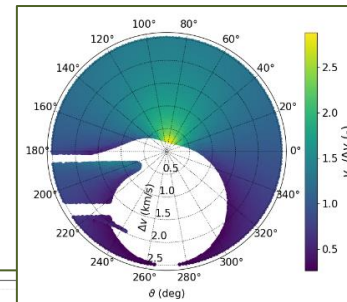
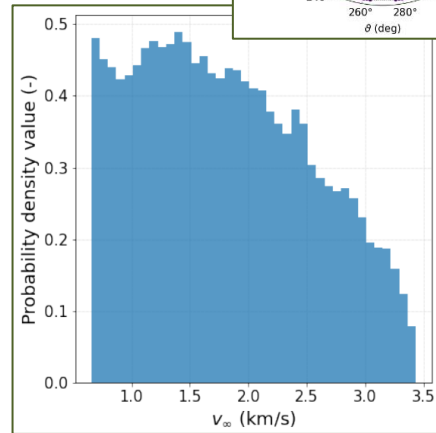
Large scale **Lambert**-based search.

Reachability Analysis

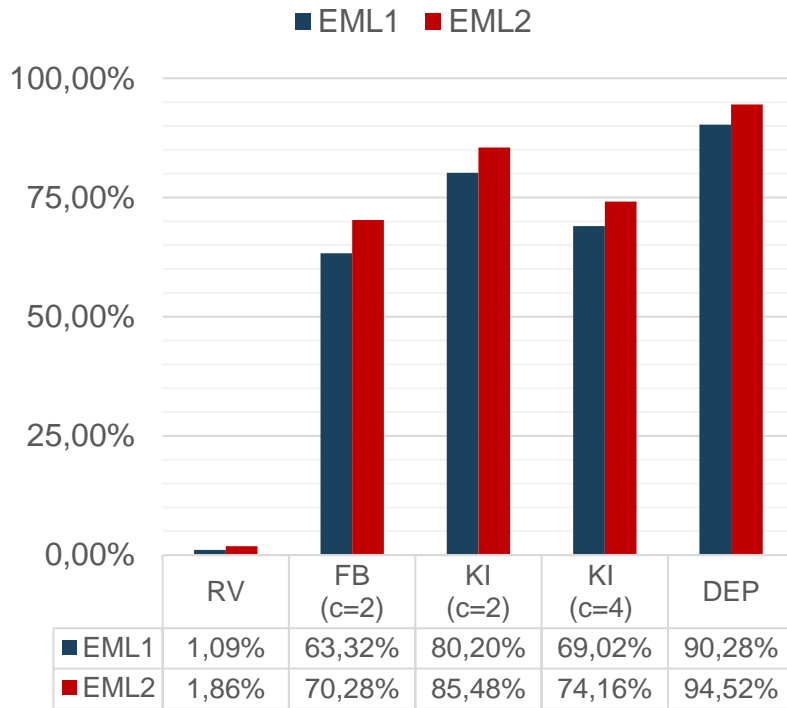
- ❑ A database of **transfers** is generated.



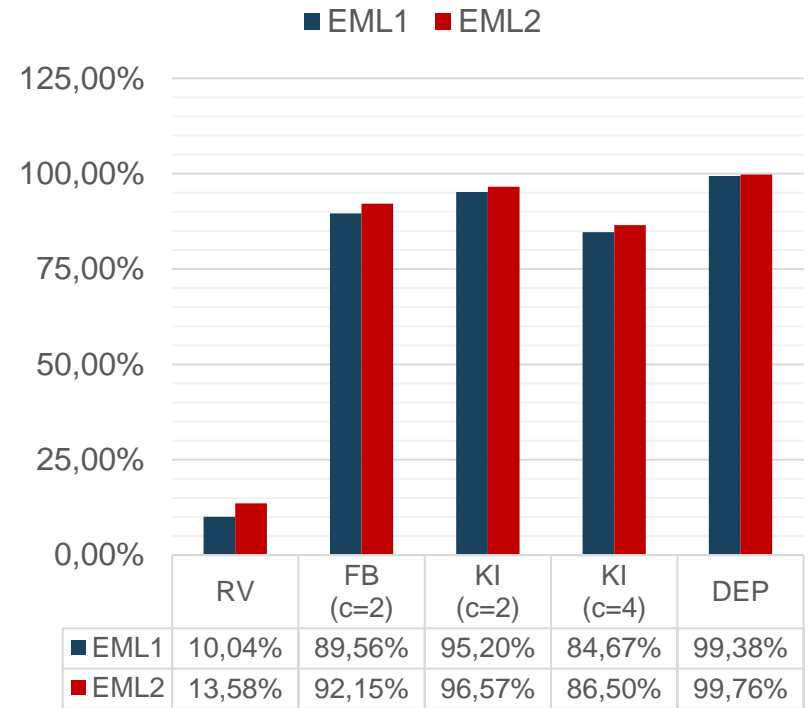
- ❑ A database of **escape conditions** is created.



Envelope Of Reachable NEA From EML1/EML2



ΔV budget of 2,5 km/s



ΔV budget of 5,0 km/s



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