

Designing the Next Generation Kinetic Impactors for Planetary Defense

Mallory E. DeCoster, Thomas Rosch, Emma Rainey, Patrick King, Angela Stickle The objective of this investigation is to provide an understanding of how the next generation kinetic impactor for planetary defense can be designed to achieve the most efficient momentum enhancement.



Illustration Credit: Henderson and Blume, Procedia Engineering 103 (2015)

Illustration Credit: NASA

Ikeda et al., Procedia Engineering 204 (2017) 138-145

The test matrix was designed to examine the effects of mass placement within the spacecraft on momentum enhancement.

Simple Impactor Geometries (2D) $\mathbf{\bullet}$ Circle Plate Ring Variables Value **Projectile Mass (g)** 1,500000 **Projectile Material** AI (2.69), Cu (8.96) (density g/cc) Impact Velocity (m/s) 500, 5000 Granite Plate: Target: Strength Strong, Medium, Weak

Complex Impactor Geometries (3D)



APL

The crater profiles for endmembers resulting from different 1g projectiles with an impact velocity of 500 m/s show that the target material properties are far more important than the impactor geometry for enhanced crater excavation.



m_{projectile}=1 g V_{impact}=500 m/s

APL

<u>**Results for 1g Projectiles:**</u> The largest craters were generated for the sphere, ogive, and plate. However crater size appears to be more dependent on the target material strength than the impactor geometry. Within similar target material parameters, the sphere emerges as the most efficient impactor geometry for optimizing beta.



<u>Results for 500 Kg Projectiles:</u> When the projectile momentum is constant, the more complex projectiles produce the largest craters. However a larger crater is not always associated with a larger beta, indicating the differences in ejecta velocity distributions may play a significant role.



<u>Summary</u>: When the impactor momentum is held constant, different projectile geometries produce varying degrees of ejecta mass resulting in ~ 38% variation in the measured momentum enhancement factor (β).





APL

nformation: JOHNS HOPKINS

Contact Information: Mallory E. DeCoster

Space and Missile Defense Applications Group Johns Hopkins University Applied Physics Lab 11100 Johns Hopkins Rd. Laurel, MD 20723-6099 Office: 240-228-2351 e-mail: <u>Mallory.decoster@jhuapl.edu</u>