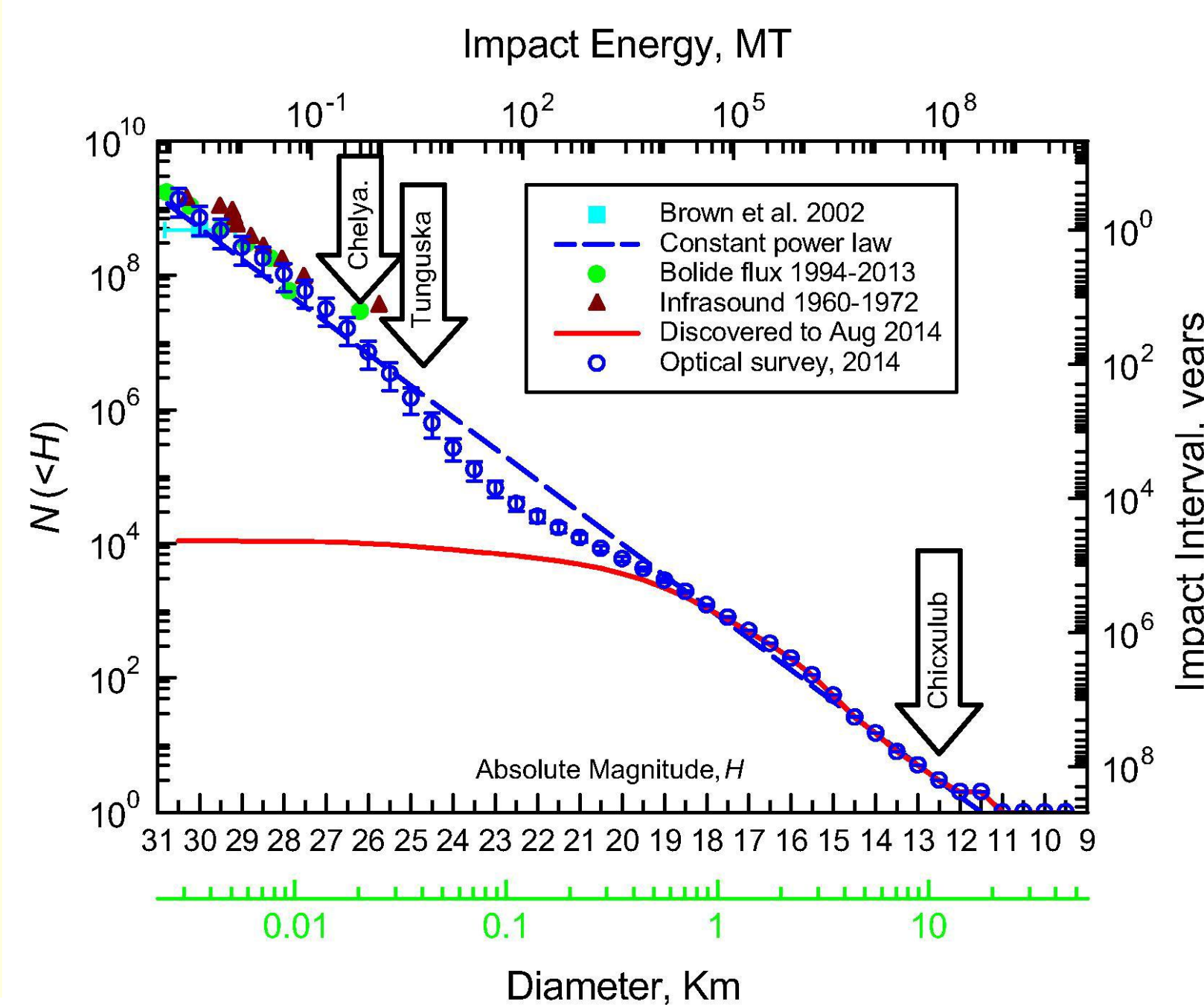
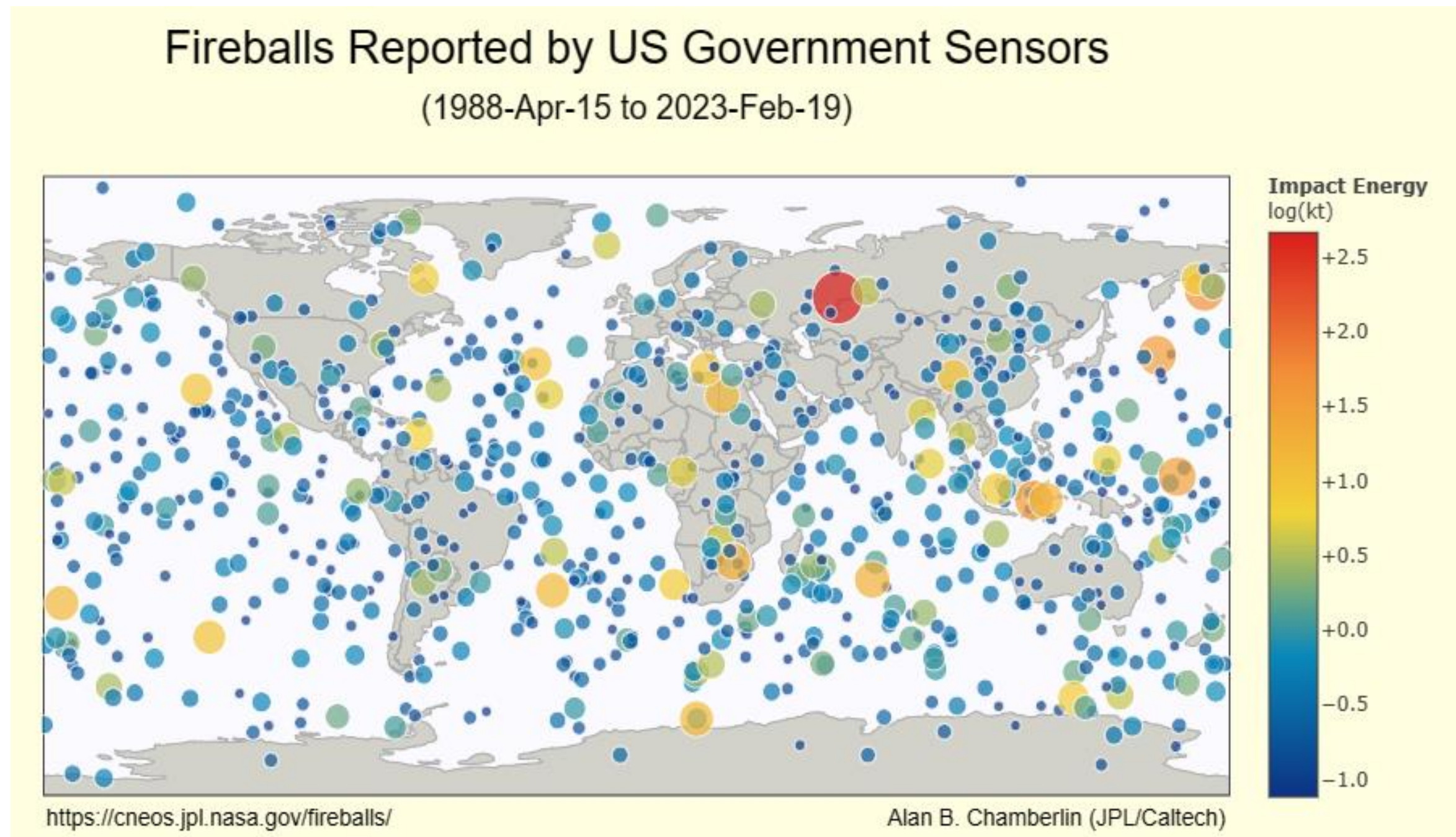


Characteristics of small-sized asteroid impacts on Earth



There are **more small-sized asteroids**; a larger proportion of them are **uncatalogued**; small-sized asteroid impacts may be **more frequent**; currently, most asteroid defense research focuses on celestial bodies of 140 meters in size, with little research on the defense of asteroids **below 50 meters** in size.

The warning time is less than 2 years
Impacting the Earth can cause a Chelyabinsk-level disaster
Nuclear explosion: prone to security issues
Kinetic Impact: can effectively defend, but it takes a long time
Kinetic destruction - effectively addressing the problem of short warning time for small-sized asteroids.

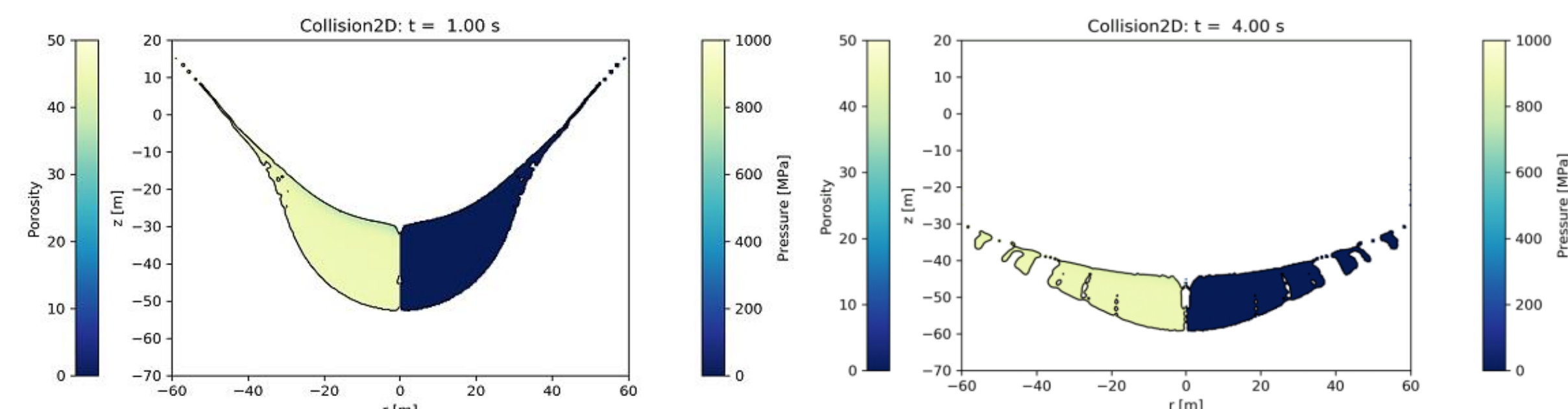
Only seven asteroid impact events (fire meteor events) have been successfully warned before they impact the Earth
The most recent impact warning: 13 February 2023, an asteroid was discovered **only a few hours** before impact

Conditions of destruction

iSALE is an impact dynamics code jointly developed by Imperial College London and the Berlin Leibniz Institute based on the SALE.

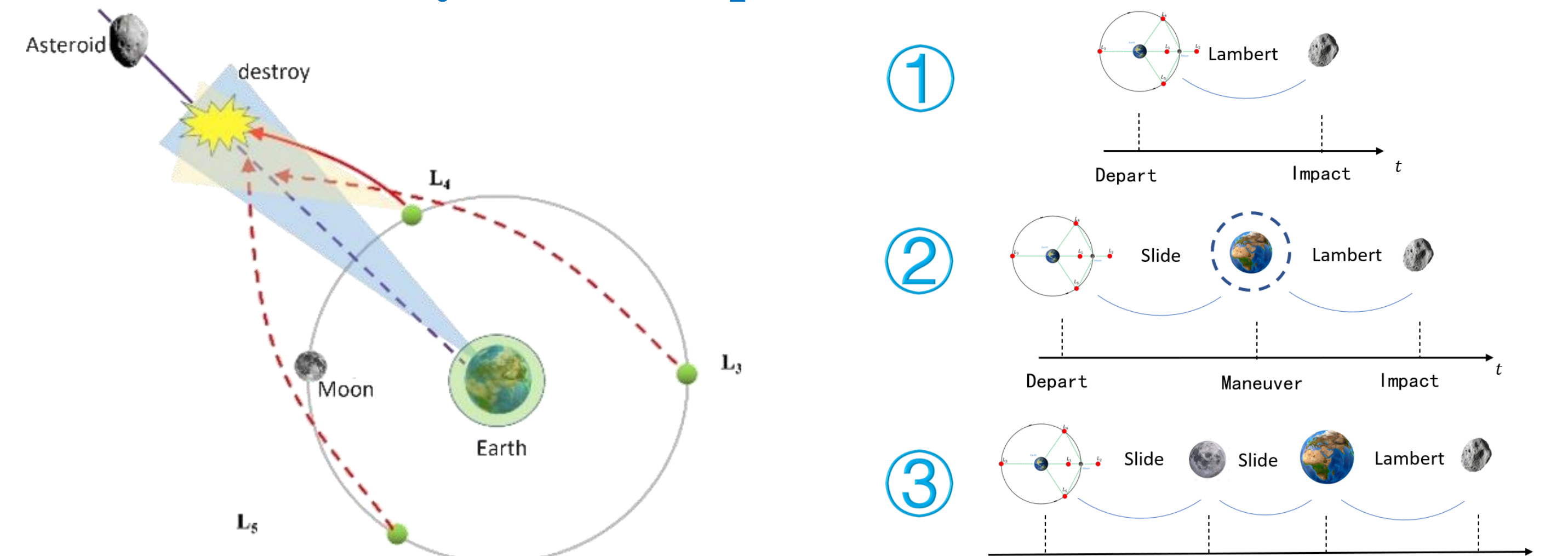
Grid type: ALE

Features : 2D+3D, Finite Difference Method, Allowing material rheology, The strength/damage model usually used is the yield stress-deviatoric strain damage model based on Collins et al. (2004), Porosity: ε - α porosity model.



A 2t impactor, impacting an S-type asteroid with a diameter of about 50 meters and a porosity of 45% at a speed of 7km/s, can destroy it.

Layout of the space-based constellation

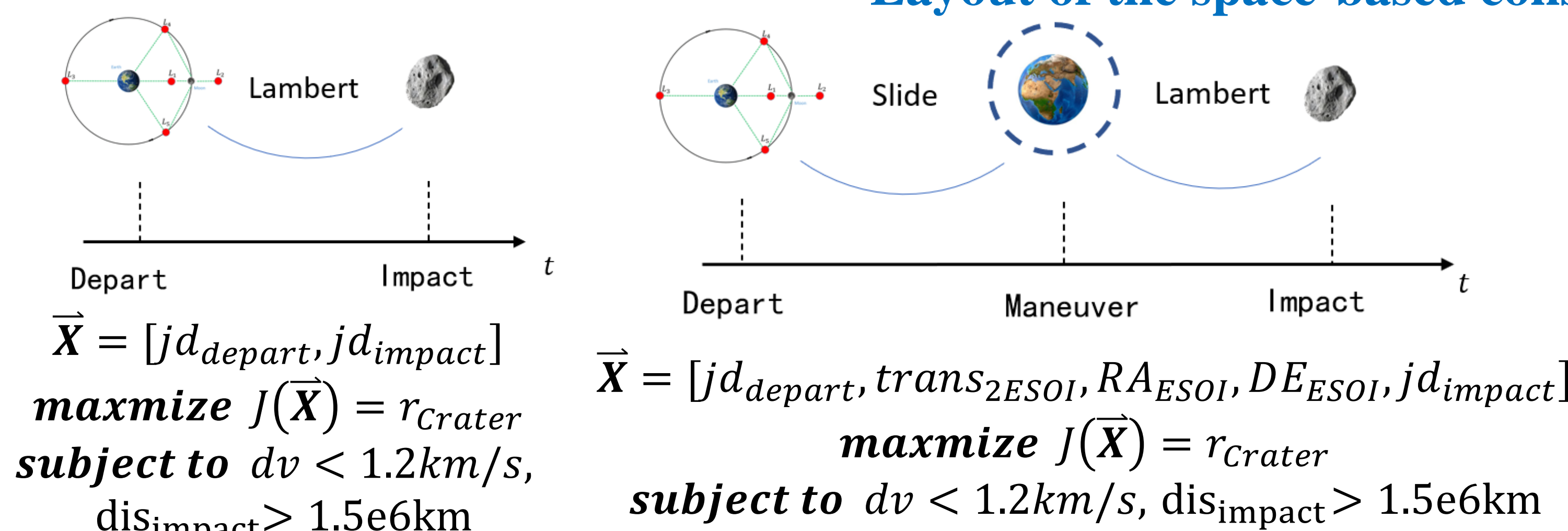


Advantages: Quickly intercept asteroids from all directions and can take into account monitoring and early warning.

$$\frac{d^2 \mathbf{r}}{dt^2} = -\frac{\mu_s}{|\mathbf{r}|^3} \mathbf{r} - \sum_{i=1}^8 \mu_{pi} \left(\frac{1}{|\mathbf{r}_{pi}|^3} \mathbf{r}_{pi} + \frac{1}{|\mathbf{p}_{pi}|^3} \mathbf{p}_{pi} \right) + \mathbf{a}_{moon} + \mathbf{a}_{GR}$$

Dynamic Model: Central body, Third body, Relativistic effects

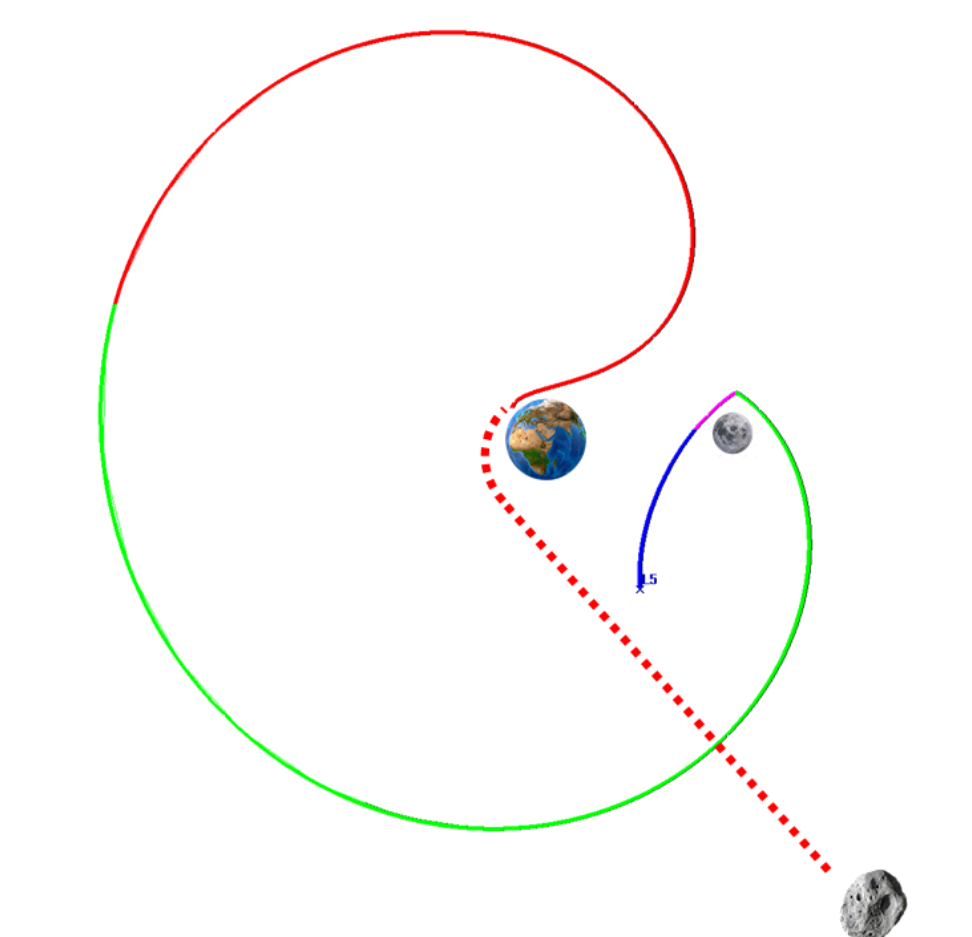
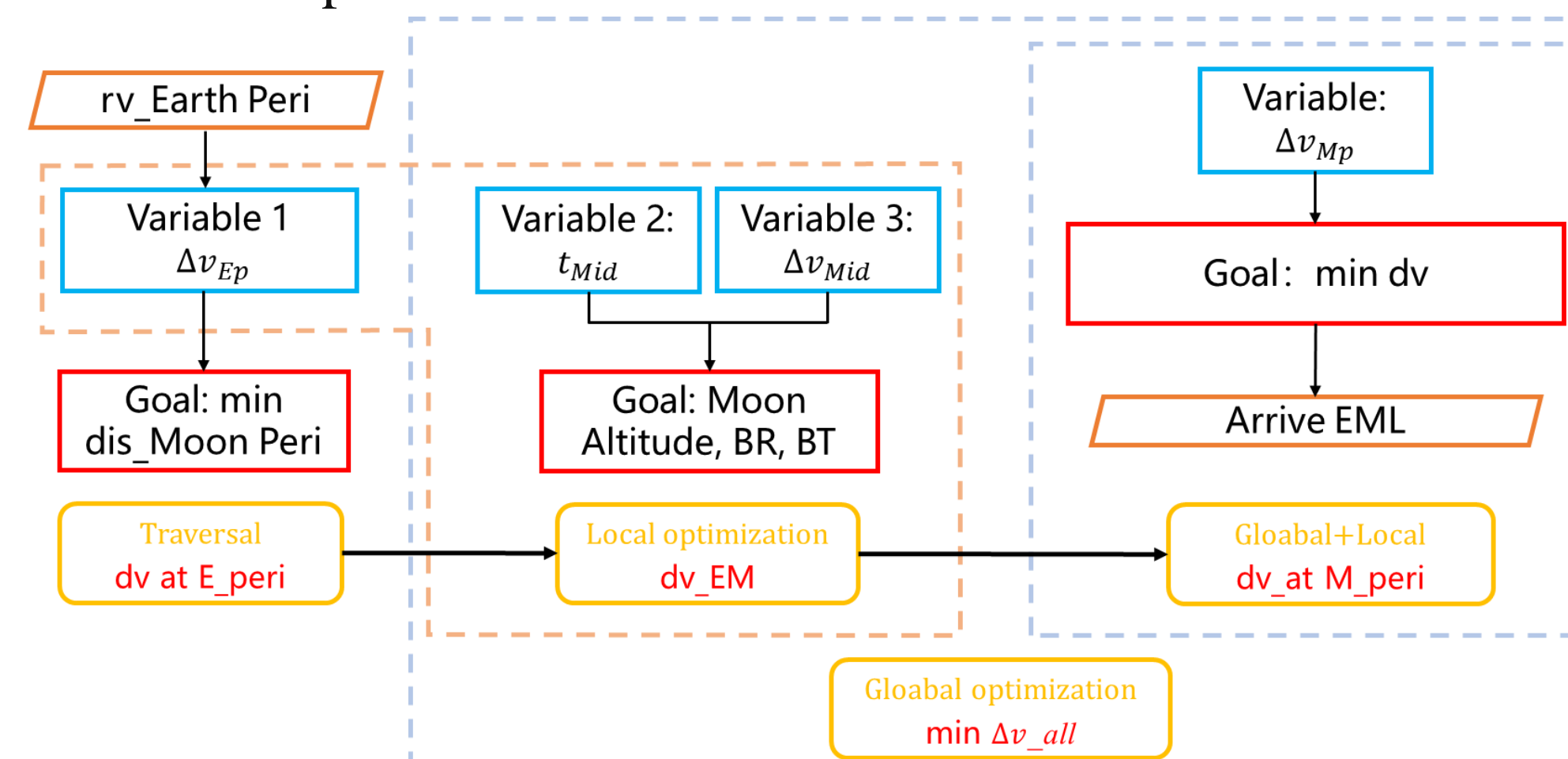
Layout of the space-based constellation—Trajectory design



| | |
|-------------|------------------|
| Asteroid | 2003 RU11 |
| Diameter | 27.26m |
| Date CA | 2034 9 17 |
| Distance CA | 3.8 E_M distance |
| Depart Date | 2034 7 27 |
| Impact Date | 2034 9 5 |
| Trans Days | 40 days |
| dv | 1.04 km/s |
| Impact v | 5.80 km/s |
| Impact mass | 714 kg |

| | |
|-------------|-------------------|
| Asteroid | 2020 UC4 |
| Diameter | 20.78m |
| Date CA | 2033 6 17 |
| Distance CA | 2.36 E_M distance |
| Depart Date | 2031 10 27 |
| Impact Date | 2032 7 20 |
| Trans Days | 268 days |
| dv | 0.52 km/s |
| Impact v | 9.04 km/s |
| Impact mass | 846 kg |

Advantages: By using the gravity assist of the Earth and Moon to change the orbital energy, the defense range can be expanded
The combination of Earth gravity assist + Moon gravity assist + long-distance transfer mode can effectively reduce fuel consumption



| | |
|---------------|-------------------|
| 小行星 | 2021 RO16 |
| 直径 | 16.88m |
| 近地日期 | 2034 4 22 |
| 近地距离 | 1.94 E_M distance |
| Depart Date | 2031 12 12 |
| Moon Ga Date | 2031 12 28 |
| Earth Ga Date | 2032 1 9 |
| Impact Date | 2034 4 15 |
| dv | 1.03 km/s |
| Impact v | 7.09 km/s |
| Impact mass | 710 kg |

| Asteroid | Diameter | cal_CA | distance |
|-----------|----------|---------------|----------|
| 2020 GB1 | 14.30 | 2035 4 7 16 | 1.35 |
| 2020 GY1 | 16.42 | 2033 4 5 5 | 2.45 |
| 2021 RO16 | 16.88 | 2034 4 22 4 | 1.94 |
| 2020 KQ4 | 17.20 | 2031 5 24 7 | 3.77 |
| 2016 JA6 | 18.86 | 2031 11 6 3 | 2.84 |
| 2014 XC8 | 18.86 | 2034 5 24 15 | 3.55 |
| 2004 DF2 | 20.68 | 2032 4 22 21 | 3.25 |
| 2020 DK | 21.06 | 2032 2 9 14 | 3.74 |
| 2020 UC4 | 20.77 | 2033 6 17 20 | 2.36 |
| 2022 CO6 | 21.16 | 2033 2 15 12 | 3.22 |
| 2011 AH5 | 21.65 | 2031 7 20 19 | 3.90 |
| 2008 DB | 23.74 | 2032 8 14 14 | 0.33 |
| 2016 DY21 | 23.74 | 2035 2 17 20 | 4.19 |
| 2014 YN | 24.86 | 2030 11 11 6 | 3.38 |
| 2020 UE | 26.76 | 2034 10 15 14 | 2.95 |
| 2003 RU11 | 27.26 | 2034 9 17 14 | 3.80 |
| 2022 SW | 32.17 | 2035 9 16 2 | 4.25 |
| 2018 VB7 | 32.77 | 2032 10 31 20 | 3.40 |
| 2019 WG2 | 37.63 | 2032 11 22 6 | 2.75 |
| 2022 QX4 | 39.58 | 2031 9 7 5 | 3.67 |
| 2020 TJ3 | 41.26 | 2035 5 3 12 | 3.88 |

Analysis of defensive effects

Select 21 asteroids as simulation targets based on the following criteria:

- real asteroids
- with a diameter of 15-50m
- Flyby Earth between 2031 and 2035
- flyby at a distance less than 5 times the Earth-Moon distance.

The impact can be completed within **two years**, with a minimum transfer time of 40 days and a maximum of no more than 630 days. The impact locations are all **outside the Earth's Sphere of Influence**. The **average velocity increment is 0.6km/s**. The maximum velocity increment **does not exceed 1.2km/s**. All asteroids can be destroyed.

| | |
|------------|---|
| Sequence 1 | All |
| Sequence 2 | 2020KQ4, 2020DK, 2020UC4, 2014YN, 2022SW, 2022QX4, 2020TJ3 (7 NEOS) |

| | dv limit | warning time | impactable asteroids | positive deflections | disruption | effective disposal | average dv |
|--------------|----------|--------------|----------------------|----------------------|------------|--------------------|------------|
| 21 Asteroids | 1.2km/s | 2years | 21 | 4 | 21 | 21 | 608.95m/s |

